

## Rules of Thumb

### TAKE OFF

- T/O distance increases 15% for each 1000' Density Altitude (DA) above sea level
- Available engine HP decreases 3% for each 1000' DA above sea level
- Fixed pitch, non-turbo a/c climb performance decreases 8% for each 1000' DA above sea level
- Variable pitch, non-turbo a/c climb performance decreases 7% for each 1000' DA above sea level
- $V_r = \sim 1.15 \times V_s$

### CLIMB

- $V_y$  decreases  $\sim 1/2$  to 1 knot for each 1000' DA
- $V_y$ ,  $V_x$ , and  $V_g$  (best glide) decreases  $\sim 1/2$  knot for each 100# under MGW
- $V_a$  decreases 1% for each 2% reduction in gross weight

### CRUISE

- TAS increases 2% over IAS for each 1000' above sea level
- Cruise fuel consumption of a non-turbo-charged a/c engine =  $1/2$  the rated HP/10

### APPROACH

- Final approach speed =  $1.3 \times V_{so}$ . This is also known as  $V_{ref}$
- A  $1^\circ$  reduction in approach angle will increase landing distance 13%
- For each 1000' increase in field elevation above sea level, stopping distance increases 4%
- $10^\circ - 25^\circ$  of flaps add more lift than drag;  $25^\circ - 40^\circ$  flaps add more drag than lift
- A tailwind of 10% of your final approach speed increases your landing distance by 20%;
- A headwind of 10% decreases landing distance by 20%
- For each knot above POH  $V_{ref}$  (final approach speed) over the runway end numbers, the touchdown point will be 100' further down the runway
- A 10% change above POH recommended final approach airspeed will cause a 20% increase in stopping distance
- A slippery or wet runway may increase your landing distance by 50% or more.